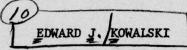


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AFPEA-77-P7-31, PTPD-77-35

TEST AND EVALUATION OF AIM 9L/MARK 17 AND 36 ROCKET MOTOR CONTAINER

AFALD/PTPD AIR FORCE PACKAGING EVALUATION AGENCY Wright-Patterson AFB OH 45433

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#### **ABSTRACT**

One each of AIM 9L/MARK 17 and 36 shipping/storage containers with inert rocket motors were received from HQ ADTC/SDMT, Eglin AFB FL at the Air Force Packaging Evaluation Agency, Wright-Patterson AFB OH. Pressure and rough handling tests were conducted on the containers in accordance with Federal Test Method Standard 101B.

Visual inspection indicated no physical damage to the containers upon completion of the tests.

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Air Force Packaging Evaluation Agency PUBLICATION DATE:

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APPROVED BY:

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Evaluation Agency

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#### INTRODUCTION

HQ ADTC/SDMT, Eglin AFB FL requested the Air Force Packaging Evaluation Agency (AFPEA), Wright-Patterson AFB OH to conduct pressure and rough handling tests on the AIM 9L/MARK 17 and 36 containers with inert rocket motors.

The AIM 9L/MARK 17 and 36 containers were originally fabricated by Metric Systems, Fort Walton Beach FL for shipment and storage of Sparrow missiles. For these tests each Sparrow container was modified to accept four inert rocket motors.

# TEST OUTLINE AND TEST EQUIPMENT

Tests were conducted in accordance with Federal Test Method Standard (FTMS) 101B, Level A packing. Figures 1 through 3 outline the container test plans.

A Meriam, Model RC-4615, water manometer graduated in 0.20 inch increments was used for the leak tests. An Endevco, Model 2233E, piezoelectric accelerometer was used to instrument all tests on the AIM 9L/MARK 17 and 36 containers. Conditioning of the accelerometer output was accomplished by an Endevco charge amplifier, Model 2641C, powered by an Endevco power supply, Model 2622C. The continuous output was displayed on a Tektronix, Model 564B, storage oscilloscope equipped with a Tektronic still camera, Model C-12.

A L.A.B. Corporation vibration machine, Serial No. 56801, Type 5000-96B which has a frequency servoloop constant displacement cam linked motor drive was used for the vibration test.

A pendulum-impact tester fabricated in accordance with Figure 1 of FTMS 101B, Method 5012, was used for the impact test.

# TEST PROCEDURES AND RESULTS

#### AIM 9L/MARK 17 CONTAINERS: The following tests were conducted:

1. LEAK TEST: Immediately upon receipt of the container at the AFPEA, a leak test was conducted on the container in accordance with the procedure in FTMS 101B, Method 5009, paragraph 1.2(c) (see Figure 4).

RESULTS: The results of this test are shown in tabulated form below:

TABLE I

TIME (SECONDS)	INCHES H <sub>2</sub> 0 DISPLACED	PSIG
00	41.60	1.50
30	41.60	1.50
60	41.60	1.50
90	41.60	1.50
120	41.60	1.50
150	41.60	1.50
180	41.60	1.50
210	41.60	1.50
240	41.60	1.50
270	41.60	1.50
300	41.60	1.50

2. VIBRATION TEST: A vibration test was conducted in accordance with the procedure in FTMS 101B, Method 5019. A one inch double amplitude and 4.5 Hz frequency was maintained for two hours (see Figure 5).

RESULTS: Visual inspection revealed no damage to the container. A maximum of 28 G's was recorded on the inert rocket motors during the test (see Figure 6).

3. CORNERWISE DROP (ROTATIONAL) TEST: The cornerwise drop (rotational) test was conducted in accordance with FTMS 101B, Method 5005. A 24 inch drop height was used during the tests. Drops were made once to each of two diagonally opposite corners of the base.

RESULTS: Visual inspection revealed that the first latch on the aft end and the first latch on the opposite end unbuckled during the first drop. A maximum of 100 G's was recorded on the inert rocket motors during the test. No damage was noted to the container.

4. EDGEWISE DROP (ROTATIONAL) TEST: The edgewise drop test was conducted in accordance with FTMS 101B, Method 5008. A 24 inch drop height was used during the tests. Drops were made once to each end of the container.

RESULTS: No damage was noted to the container. A maximum of 150 G's was recorded on the inert rocket motors during the test.

5. PENDULUM-IMPACT TEST: The pendulum-impact test was conducted in accordance with FTMS 101B, Method 5012. Impact was at seven feet per second. Both ends and both sides were impacted.

RESULTS: Visual inspection revealed that a pop rivet on the gasket retaining aluminum strip on the aft end sheared during the impact tests. A maximum of 12 G's was recorded on the inert rocket motors during the tests. No other damage was noted to the container.

6. LEAK TEST: At the end of the rough handling tests the container was leak tested in accordance with the procedure in FTMS 101B, Method 5009, paragraph 1.2(c).

RESULTS: Air pressure could not be maintained. Leaks were noted on the front end, at the spot welds (see Figure 7), and at the joint of the gasket.

# AIM 9L/MARK 36 CONTAINER: The following tests were conducted:

1. LEAK TEST: Immediately upon receipt of the container at the AFPEA, a leak test was conducted on the container in accordance with the procedure in FTMS 101B, Method 5009, paragraph 1.2(c).

RESULTS: The results of this test are shown in tabulated form below:

#### TABLE II

TIME (SECONDS)	INCHES H20 DISPLACED	PSIG
00	41.45	1.498
30	41.45	1.498
60	41.45	1.498
90	41.45	1.498

TABLE II (Continued)

TIME (SECONDS)	INCHES H20 DISPLACED	PSIG
120	41.43	1.497
150	41.42	1.496
180	41.42	1.496
210	41.42	1.496
240	41.41	1.496
270	41.40	1.495
300	41.40	1.495

2. VIBRATION TEST: A vibration test was conducted in accordance with the procedure in FTMS 101B, Method 5019. A one inch double amplitude and 4.5 Hz frequency was maintained for two hours.

RESULTS: Visual inspection revealed that the pin which holds the latch bolt to the handle dropped out during the test and the cushioning in the top of the container slipped from center to the aft end (from 34 to 18 1/4 inches). A maximum of 5 G's was recorded on the inert rocket motors during the test.

3. CORNERWISE DROP (ROTATIONAL) TEST: The cornerwise drop test was conducted in accordance with FTMS 101B, Method 5005. A 24 inch drop height was used during the tests. Drops were made once to each of two diagonally opposite corners of the base.

RESULTS: Visual inspection revealed no damage to the containers. A maximum of 18 G's was recorded on the inert rocket motors during the test.

4. EDGEWISE DROP (ROTATIONAL) TEST: The edgewise drop test was conducted in accordance with FTMS 101B, Method 5008. A 24 inch drop height was used during the test. Drops were made once to each end of the container.

RESULTS: Visual inspection indicated no damage to the container. A maximum of 14 G's was recorded on the inert rocket motors during the tests.

5. PENDULUM-IMPACT TEST: The pendulum-impact test was conducted in accordance with FTMS 101B, Method 5012. Impact was at seven feet per second. Both ends and both sides were impacted.

RESULTS: Visual inspection revealed that the fin mounts of one motor made contact with another motor during the side impact. No readings were recorded. A maximum of 26 G's was recorded from end impact, on the inert rocket motors during the tests.

6. LEAK TEST: At the end of the rough handling tests the container was leak tested in accordance with FTMS 101B, Method 5009, paragraph 1.2(c).

RESULTS: Air pressure could not be maintained. Leaks were noted on front end, at spot welds, and at the joint of the gasket.

Test plan, Figure 3, was conducted on the AIM 9L/MARK 36 container the redesigned cushioning was received from Eglin AFB FL. Tests again conducted in accordance with FTMS 101B.

RESULTS: Visual inspection indicated no damage to the container. Maximum G readings recorded on the inert rocket motors during the tests were as follows:

- a. Vibration Test 5 G's
- b. Cornerwise Drop (Rotational) Test 20 G's
- c. Edgewise Drop (Rotational) Test 20 G's
- d. Pendulum-Impact Test 15 G's

#### DISCUSSION

Visual inspection of the containers at the end of the rough handling tests indicated no physical damage to the containers.

Instrumentation indicates the AIM 9L/MARK 36 has better designed cushioning than the AIM-9L/MARK 17 container.

	AIR FO		CKAGING entainer Te			NCY	77-P7-31	CT NUMBER
CONTAINER SI	ZE	(GROSS)	(ITEM)	CUBE	101	QUANTITY	DATE	7-81/2" 38-1
5-1/2×18-	1/2-23	553	345	21.09		1	2 A	ug 77
TEM NAME	dought at	gleW ee	ol . metalika	a all y rest	MANUFAC		ladyoll track	
IM-9L/MK	17/INERT	ROCKET	MOTOR	Park reco		Systems, For	t Walton Bea	ich, FL
CONTAINER NA					\$150 (A			APPEN EX
/L (AFPE		ition)			19130 (Y	pprox)		<del>-</del> _
PACK DESCRIP Metal top		nam erdeh	0001 000	rot				
		OII WILL	sear gas	NGC .				
Ambient or		cribed	by test			2092 70	hedimasın	65 28 975136
TEST NO.	1	AW	PAR	AMETERS	2000 (310)	ORIENT	ATION	INSTRUMENTE
LEAK TEST								1271 461
1	FTMS :		of Lenny			1.8.9.0.5	Hate Section	
		1 5009	2.0 P.	S. I.		Normal Posi	tion	Manometer
	Para .	1.2 (C)						ET SMITTAGET
VIBRATION	TEST							
2	FTMS	101	1" Dou	ble AMP.	1200	As required	by test	Inst.
	Metho	d 5019		, 2 Hours				
								K RESTAN SERVICE
ROUGH HAN	DLING TES	STS						
3	FTMS :	101	24 inc	h Drop He	ight	Once to each	h of two	Inst.
	Metho	d 5005	in constant			Diagonally		
		tone da				corners of	base	
4	FTMS :	101	24 inc	h Drop He	ight	Once to eac	h end of	Inst.
		d 5008				container		
791	94	Ta door	minte dros		150	day sail 1	\$0.0 delega	
5	FTMS	101 d 5012	7 FPS	Impact		Both ends -	Both sides	Inst.
	Metho	1 3012						
LEAK TEST								
6	FTMS	101 d 5009	2.0 P.	S T		Normal Posi	tion	Manometer
		1.2 (C)	2.0 1.	J. 1.		NOTHIAL TOST	LION	nationecci
		(0)						
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		(C	ontainer Tes	t Plan)			77-P7-3	
CONTAINER SIZE	11-1	(GROSS)	WT (ITEM)	CUBE		QUANTITY	DATE	
77-3/4"x18-1/	2"x23	572	384 1/2	19	.20	1	2 /	Aug 77
ITEM NAME					MANUFA	CTURER	6.0	
AIM-9L/MK-36	Inert	Rocket	Motors		Metri	c Systems, Fo	rt Walton Be	each, FL
CONTAINER NAME					CONTAI	NER COST		
C/S (AFPEA	Desig	nation)			\$150	(Approx)		1300 0000
PACK DESCRIPTION	N							
Metal, top ar	d bot	tom wit	h seal gas	ket				THE CALLERY
Ambient or as	pres	cribed	by test		٠			39136773
TEST NO.	•	AW	P/	ARAMETERS		ORIENT	ATION	INSTRUMENT
LEAK TEST		101 od 5009 1.2 (0		.1.		Normal Pos	ition	Manometer
VIBRATION TES	<u>T</u>							
2	FTMS Metho	101 od 5019	1" Doub.	le AMP.		As required	d by test	Inst
ROUGH HANDLIN	G TES	<u>rs</u>						
3	FTMS Meth	101 od 5005	24 inch	drop hei	ght	Once to eac diagonally corners of	opposite	Inst
4		101 od 5008		drop hei	gḥt	Once to eacontainer	ch end of	Inst
5	FTMS	101 od 5012	7 FPS I	mpact		Both ends-	both sides	Inst
LEAK TEST								
6		101 od 5009 1.2 (0		.1.				Manometer
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	AIR F		CKAGIN ontainer		VALUATION Plan)	ON AGEN	ICY		EA PROJI	ECT NUMBER	
CONTAINER SIZE		(GROSS)		TEM)	CUBE	19.20	QUANTITY 1		DATE 15	Aug 77	
TEM NAME AIM-9L/MK-36	Iner	Rocket	Motors			MANUFA Met	TURER ric System	s, Fort W	L		L L
CONTAINER NAME		nation)				CONTAIN	ER COST \$15	0 (Approx	r)		_
PACK DESCRIPTION		al top a	and bott	om v	with seal	gasket					_
CONDITIONING											
TEST NO.		IAW		PA	RAMETERS		ORI	ENTATION		INSTRUME	NT
IBRATION TEST 2		3 101 nod 5019			AMP. 2 Hours		As requ	ired by t	est	Inst	
OUGH ANDLING EST 3		5 101 nod 5005		ch d	lrop heig	ght	diagona	each of 11y oppos of base		Inst	
4	1000	5 101 nod 5008	24 in	ch d	irop heig	sht	Once to	each end	lof	Inst	
5		5 101 nod 5012	7 FPS	Imp	pact		Both en	ds-both s	ides	Inst	
							—-е	00RDI	NATL	D N	_
COMMENTS							SYMBO SDMI FALD PT PT		ME Medde endue	DATE /&/Aug/7 //GOUL	
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FIGURE 3

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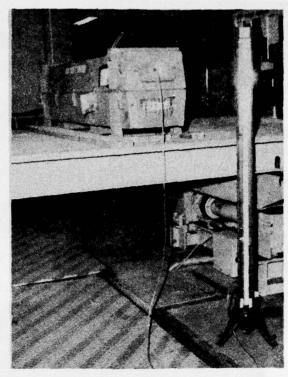


FIGURE 4. LEAK TEST USING WATER MANOMETER

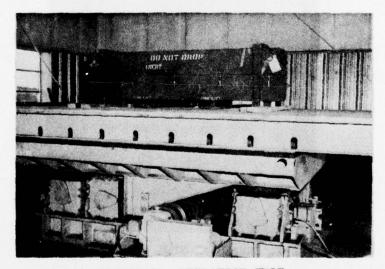


FIGURE 5. VIBRATION TEST



FIGURE 6. ACCELEROMETER MOUNTED ON ROCKET MOTOR

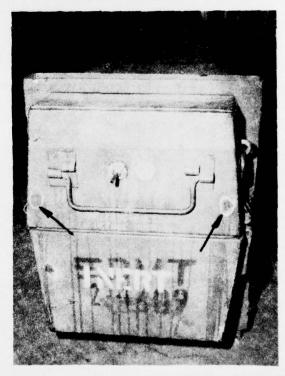


FIGURE 7. LEAKS AT SPOT WELDS, FRONT END

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ROCKET MOTOR		
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Evaluation Agency, Wright-Patters	on AFB OH. Pressu	re and rough handling tests
were conducted on the containers		
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Visual inspection indicated no physical damage to the containers upon completion

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